



Numerical Modeling of Leading-Edge Ceramic Oxidation in Inductively Coupled Plasma Facility

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CHES
CENTER FOR HYPERSONICS
&
ENTRY SYSTEMS STUDIES

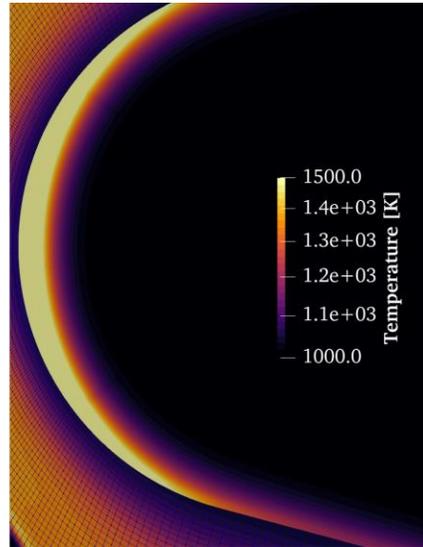
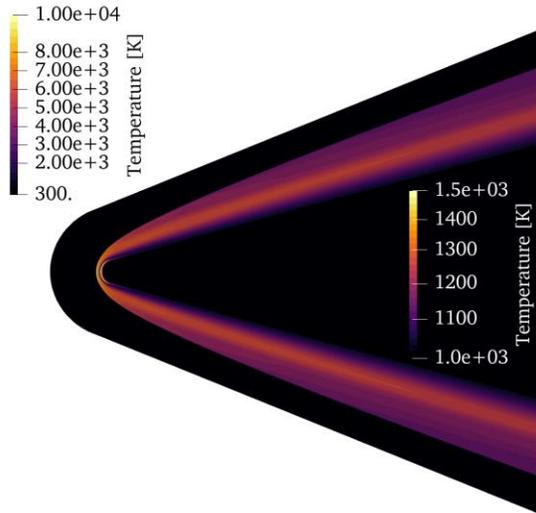


Ablation Workshop
November 9th-10th 2022

Ceramics Usage for Hypersonic Applications



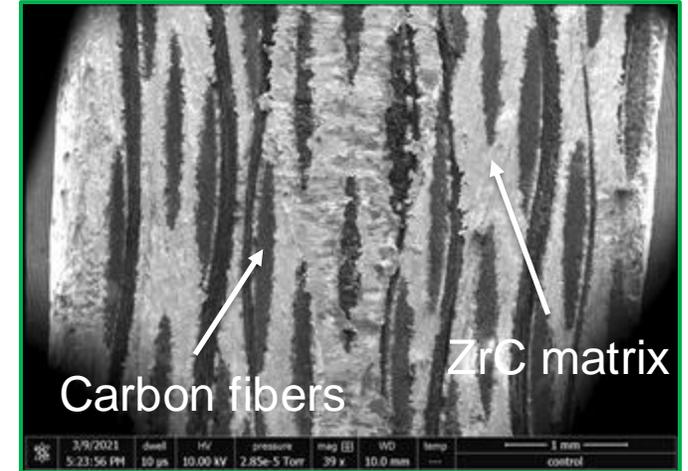
Thermal Protection Systems for atmospheric reentry necessitates high temperature and stress resistance.



Hypersonic reentry simulation at Mach 26.



C/ZrC ceramics composite



Scanning Electron Microscopy

Furnace experiment in an oxygen rich environment



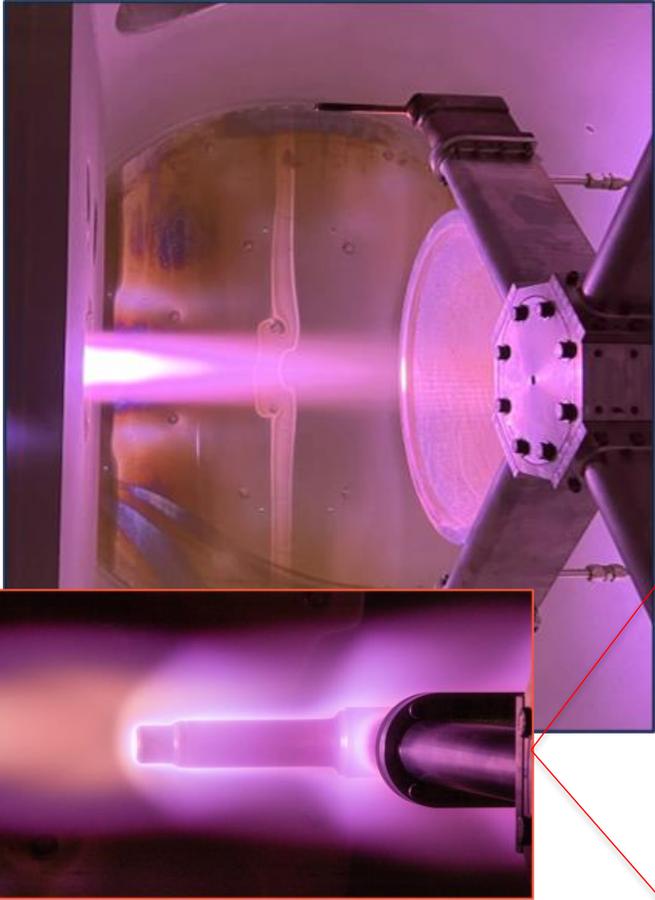
Medium temperature : powdered oxide layer



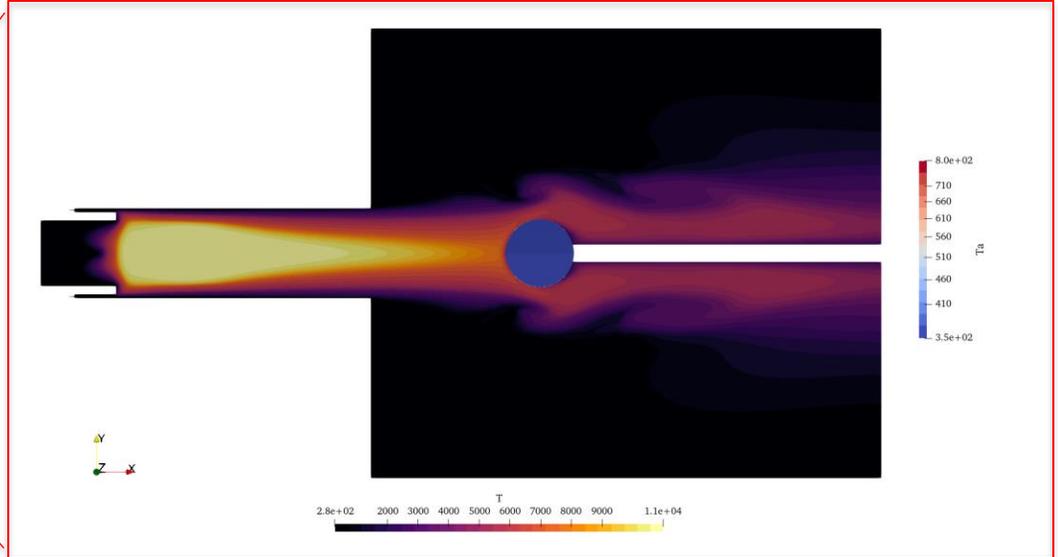
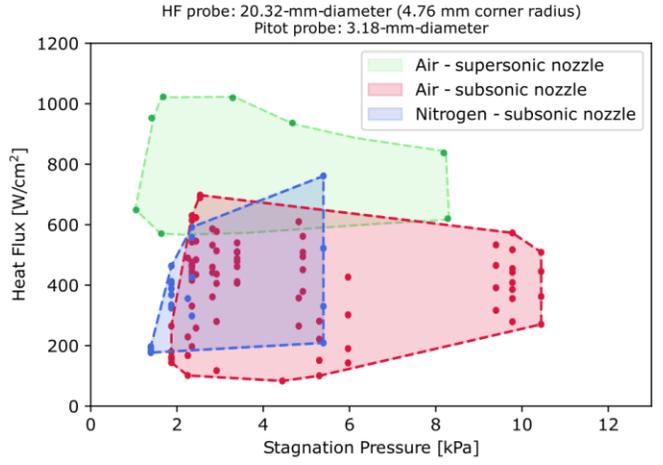
High temperature : dense oxide layer



Thermal Protection System Testing



- Inductively Coupled Plasma (ICP) technology
- 350 kW Radio-Frequency Generator (2.1 MHz)
- Maximum performances:
 - Mach 4
 - Stagnation Enthalpies up to 36 MJ/kg
 - Pressure up to 5 bar



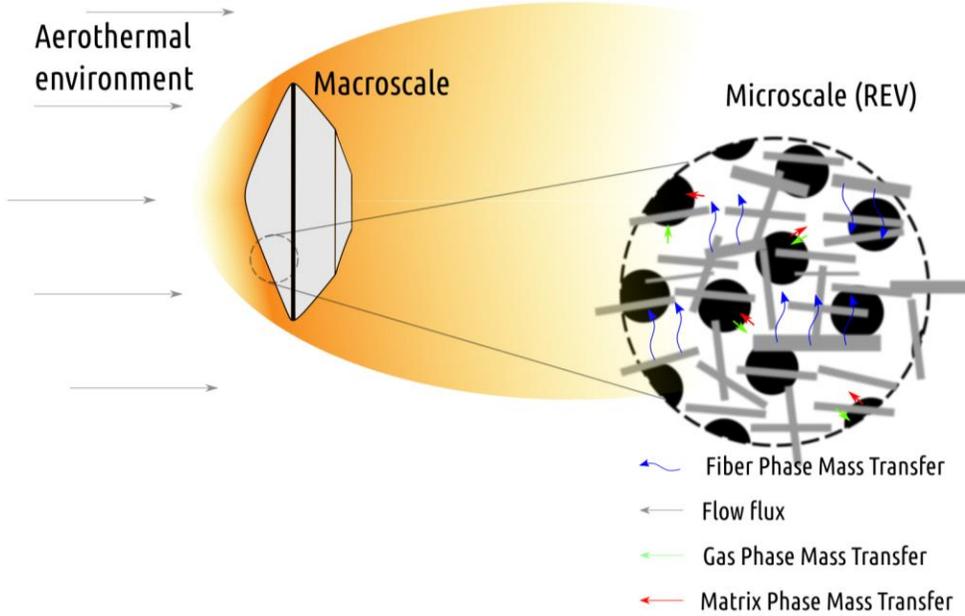
Specification overview of some materials testing facilities



Digital twin of ICP experiment (Non-Equilibrium Flow) 2D axisymmetric



CMC Thermophysical Model



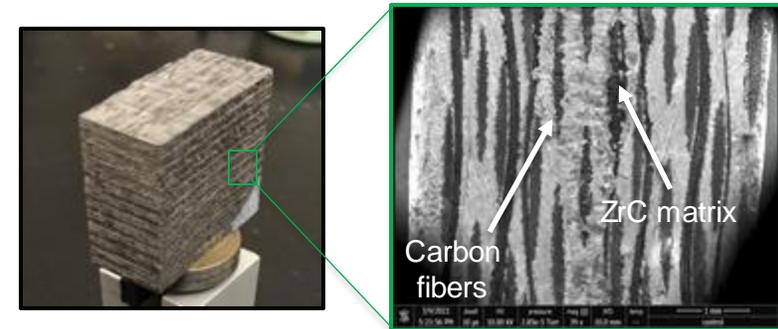
Thermal Protection Systems for atmospheric reentry necessitates high temperature and stress resistance.

The **design of protection material** is critical to assess the **safety** of the **payload**.

Simulation of Material Response in Aerothermal Environment necessitates :

- Ablation arbitrary number of solid phases,
- Heterogeneous mass exchange,
- Gas species diffusion through the material,
- Global mass and energy conservation.

Spatial resolution
Low High



High Low
Domain integration

How to leverage the loss of information on large scale modelling ?



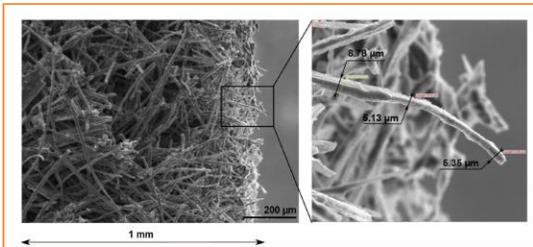
Multiscale Description of TPS

Detailed microscopic mass, momentum and energy balance at the scale for the Representative Elementary Element (REV)

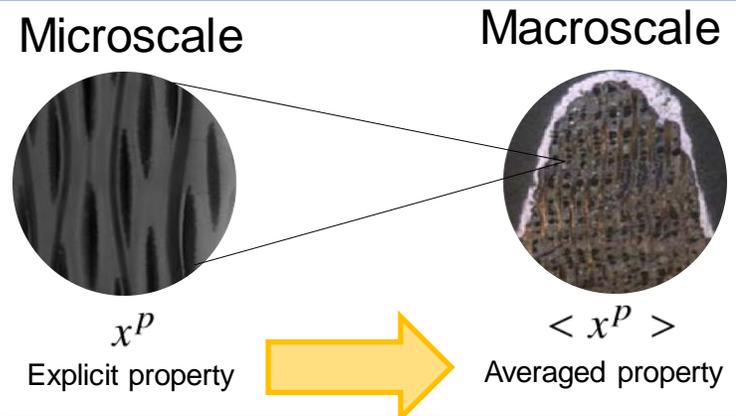
$$\langle x^p \rangle = \frac{1}{V_{REV}} \int_{D^p} x^p dV \quad \forall p \in \mathcal{P}$$

Mathematical Average of microscale governing equations on the REV

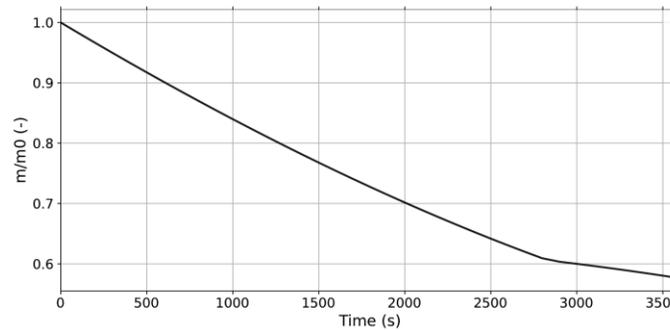
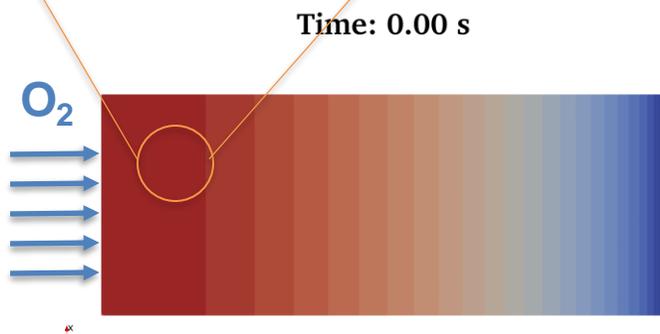
Multiscale ablation of carbon fibers



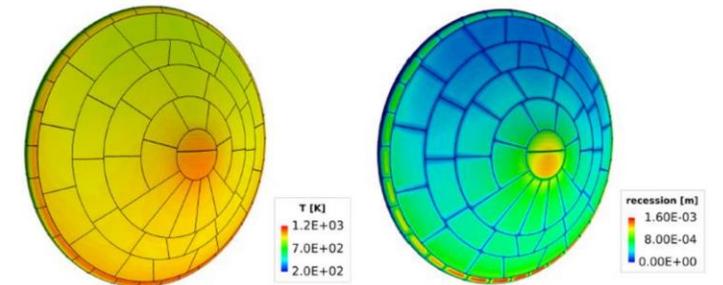
Cylindrical Shrinkage Fiber Model



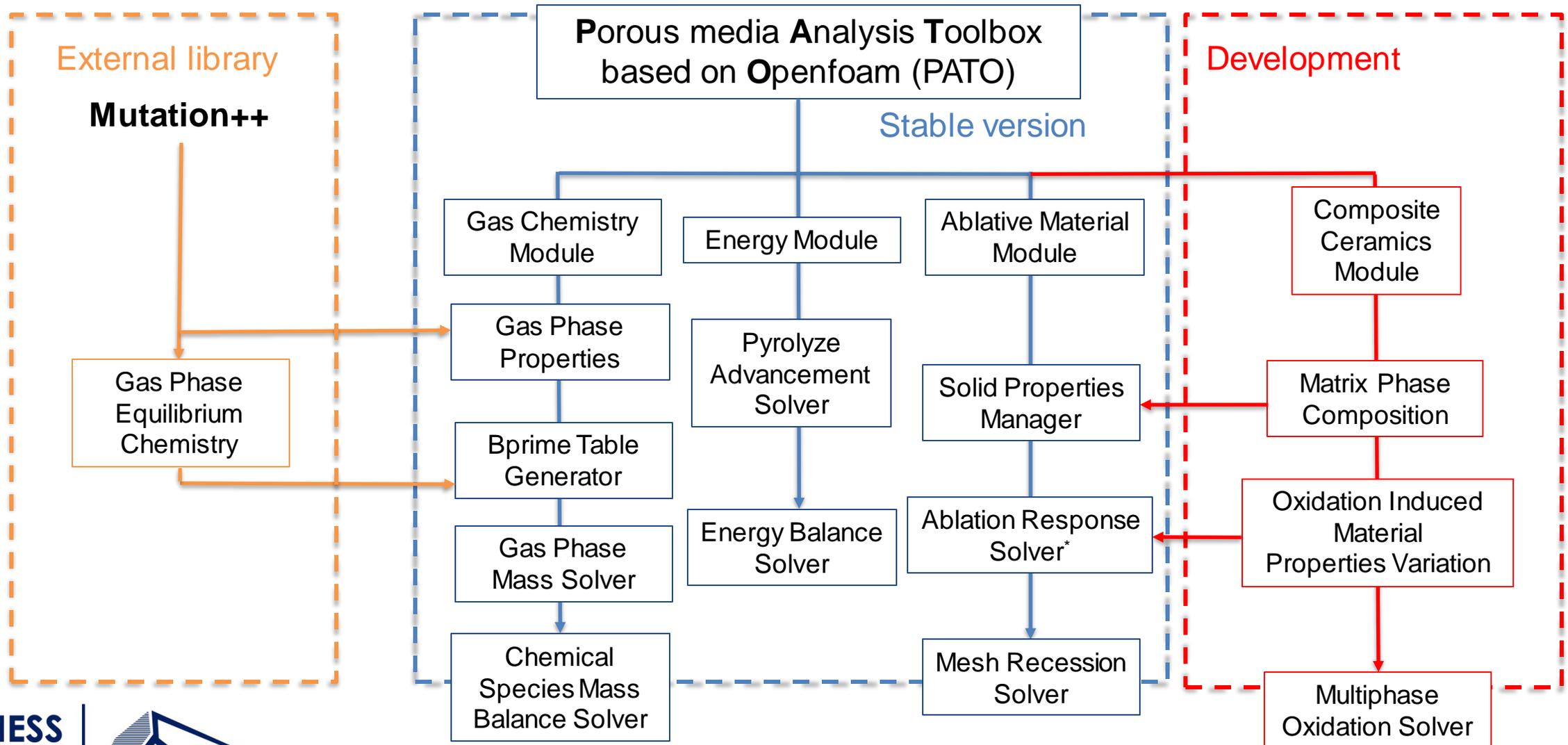
Numerical approximation of resulting PDE on the domain of interest



e.g., Thermochemical simulation of MSL heatshield [1]:



PATO code overview



*Concern only carbon fiber shrinkage module



Mathematical Model for Reactive Porous Ceramics

Mass Conservation in the Gas phase :

- The perfect gas law is assumed for the gas phase $\bar{\rho}^g = \frac{\bar{p}^g RT^g}{M^g}$
- Velocity of gas is computed using the Darcy-Klinkenberg law $\bar{v}^g = -\frac{1}{\epsilon^g} \left(\frac{K^g}{\mu^g} - \frac{\beta}{\bar{p}^g} \right) \nabla \bar{p}^g$
- This allow to define the mass conservation equation of the phase solely as a function of pressure :

$$\frac{\partial}{\partial t} \left(\epsilon^g \frac{\bar{p}^g RT^g}{M^g} \right) - \nabla \cdot \left(\frac{\bar{p}^g RT^g}{M^g} \left(\frac{K^g}{\mu^g} - \frac{\beta}{\bar{p}^g} \right) \nabla \bar{p}^g \right) = \sum_{ns=1}^{N_s} \left(\theta^{ns} \sum_k^{I_g} \dot{\omega}_k^{g,ns} \right)$$

Heterogeneous reactions
for all solid phases

- And for gas species k :

$$\frac{\partial}{\partial t} \left(\epsilon^g (\rho^g y_k^g) \bar{g} \right) + \nabla \cdot \left(\epsilon^g (\rho^g y_k^g) \bar{g} \bar{v}^g \right) - \nabla \cdot \left(D_{k \text{ eff}}^g \nabla (\rho^g y_k^g) \bar{g} \right) = \underbrace{\epsilon^g \dot{\omega}_k^{\bar{g}}}_{\text{Homogeneous reactions}} + \sum_{ns=1}^{N_s} \underbrace{(\theta^{ns} \dot{\omega}_k^{g,ns})}_{\text{Heterogeneous reactions}}$$

Diffusion in porous
media

Homogeneous
reactions

Heterogeneous
reactions





PATO Oxidation Solver Extension

Accounting for multiphase oxidations necessitates to extend heterogenous reactions solver to keep track of gas/surface interactions and their effect on solid mass balance.

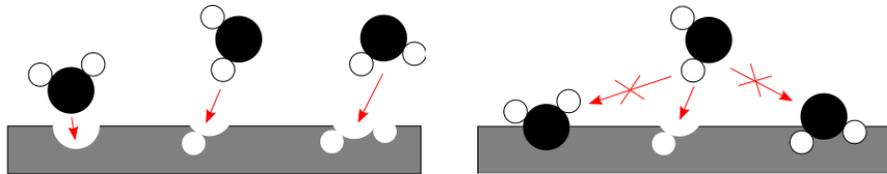
Depths of O₂ penetration is a competitive process between diffusion and surface reactivity measured by the number *Damköhler*:

Oxidation of CMC

→ O₂ flux ■ Virgin CMC □ Oxidized CMC

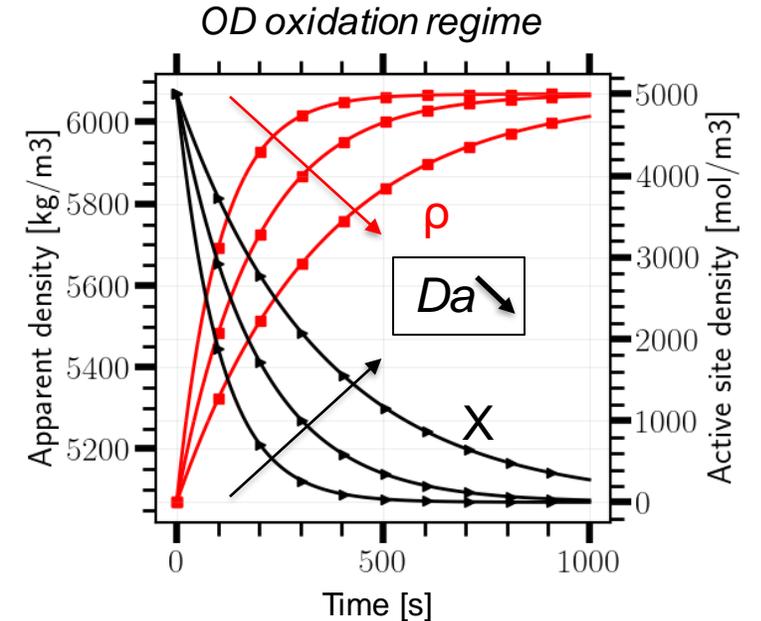
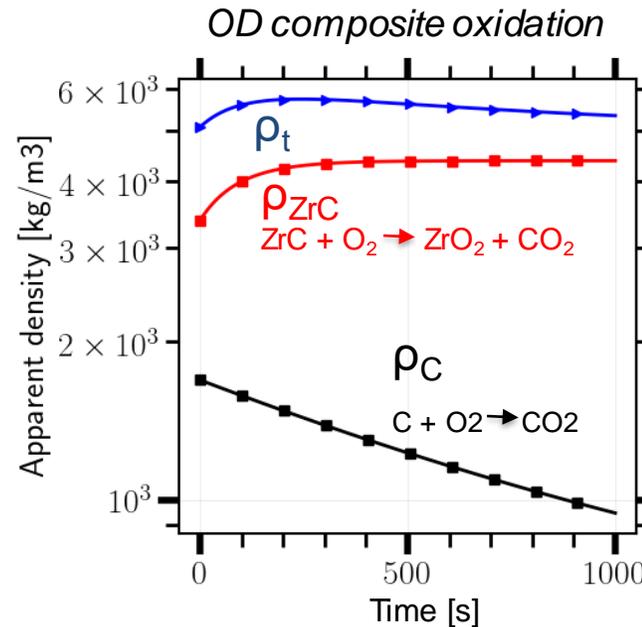


Example : adsorption on a surface



Heterogeneous source term :

$$\frac{\dot{\omega}_k^{g,ns}}{M_k} = \sum_{r=1}^{N_r} \left(\underbrace{k_r^f (\nu''_{r,k} - \nu'_{r,k}) \prod_{j=1}^{I_g \cup I_{ns}} (x_j^{g,ns})^{\nu'_{r,j}}}_{\text{mass production by forward reaction}} - \underbrace{k_r^b (\nu''_{r,k} - \nu'_{r,k}) \prod_{j=1}^{I_g \cup I_{ns}} (x_j^{g,ns})^{\nu''_{r,j}}}_{\text{mass destruction by backward reaction}} \right)$$





Energy Conservation in the porous media :

- Thermal Equilibrium is supposed to hold in the entire domain $T^g = T^{ns} \quad \forall ns \in [1, N_s]$
- The conduction parameter is an average of gas and solid phases thermal properties

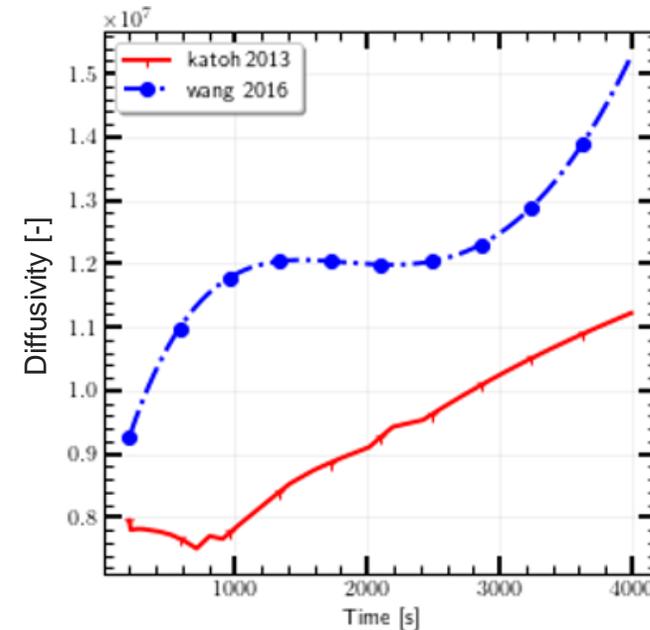
$$\sum_{ns=1}^{N_s} \underbrace{(\epsilon^{ns} \rho^{ns} c_p^{ns})}_{\text{Temperature time variation (heat storage) in solid phases}} \partial_t T + \underbrace{\partial_t (\epsilon^g \rho^g h^g - \epsilon^g p^g)}_{\text{Energy time variation in gas phase}} - \underbrace{\nabla \cdot (\lambda \nabla T)}_{\text{Heat conduction}} + \underbrace{\nabla \cdot (\epsilon^g \rho^g h^g \mathbf{v}^g)}_{\text{Heat convection}} - \underbrace{\nabla \cdot \sum_k^{I_g} (D_{k,\text{eff}}^g \nabla (\rho^g y_k^g) \bar{g} h_k^g)}_{\text{Heat diffusion}} = \underbrace{\dot{E}_\Omega + \bar{h} \pi_{\text{tot}}}_{\text{Heat source}}$$

Material and Gas Phase Properties :

- Material parameters are function of pressure, temperature and pyrolysis rate
- Gas transport properties are computed using Mutation++

Wei, X. *et al.*, Zirconium Carbide Produced by Spark Plasma Sintering and Hot Pressing: Densification Kinetics, Grain Growth, and Thermal Properties. *Materials* **2016**, 9, 577.

Katoh, Y. *et al.*, Properties of zirconium carbide for nuclear fuel applications. *Journal of Nuclear Materials*, **2013**, 741, 1.

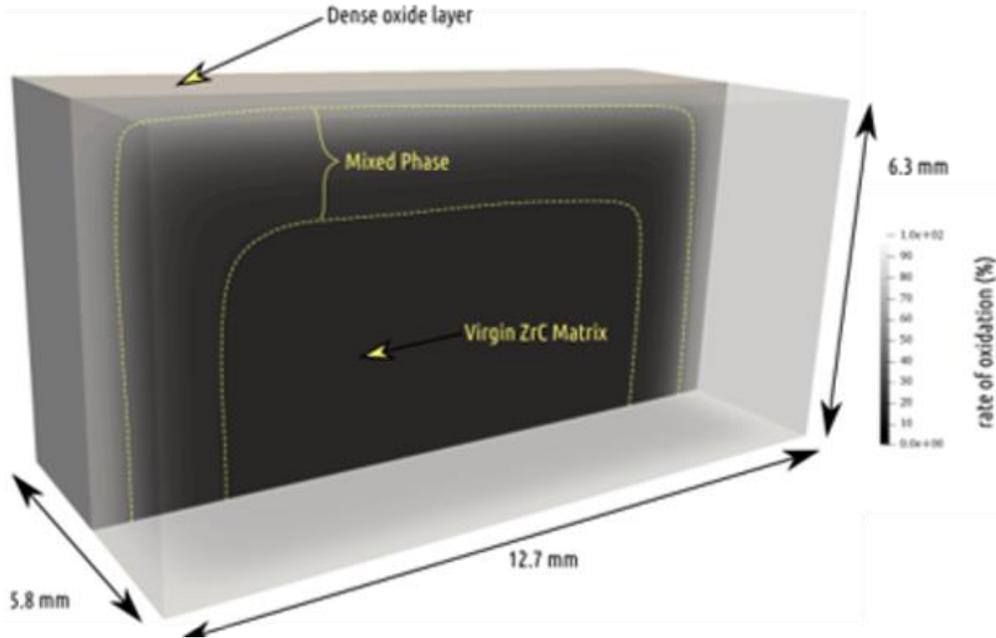




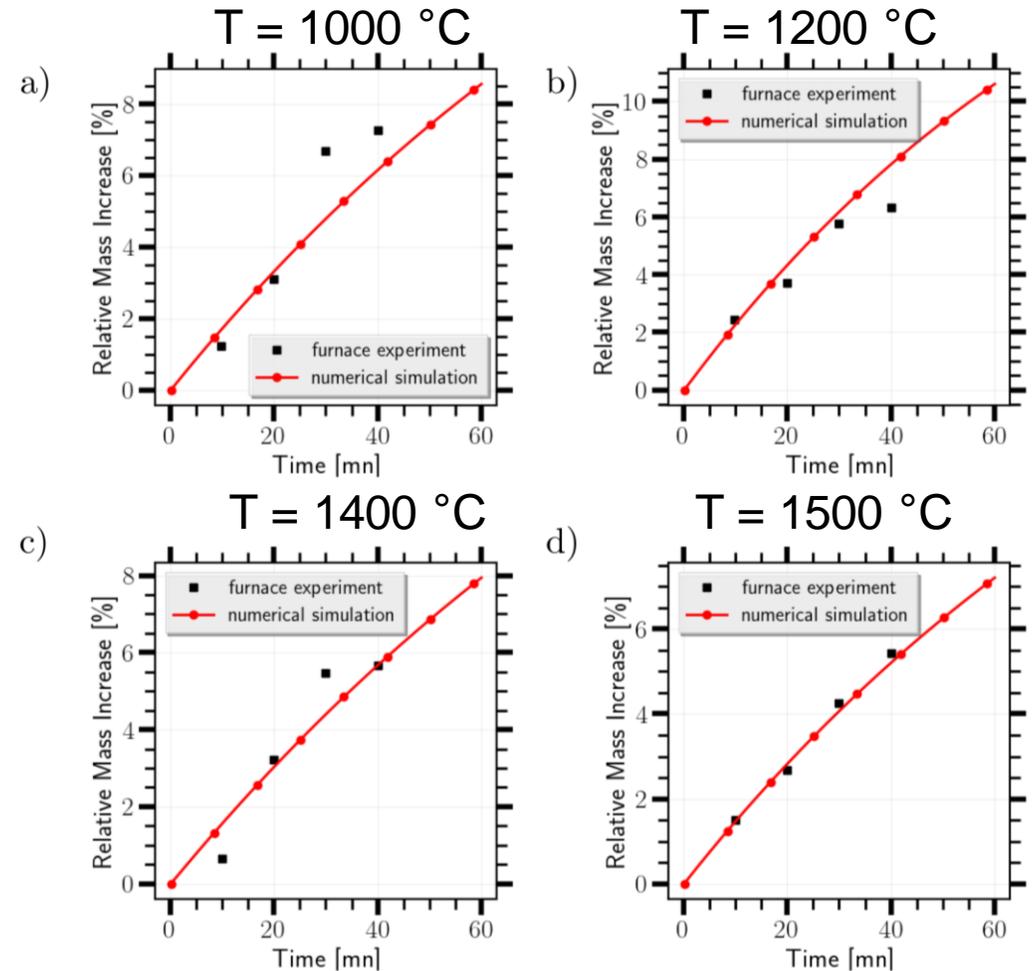
Multidimensional Oxidation

The model is applied to predict the mass variation of pure ZrC ceramics exposed to constant high temperature at atmospheric pressure environment.

Oxidation of ZrC ceramics at 1400 °C after 20mn

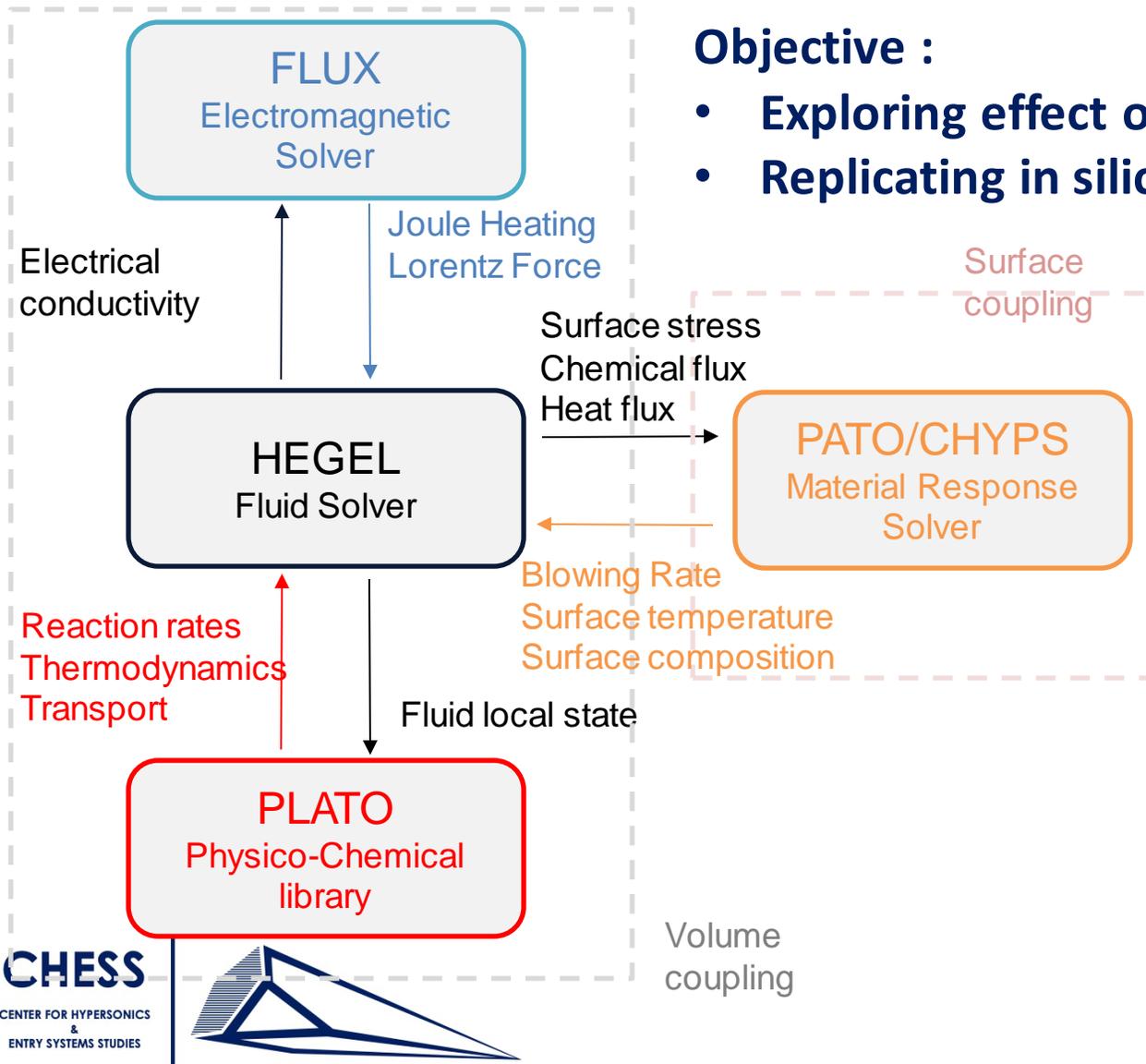


Kinetics rate are obtained from re-characterization of Rama Rao oxidation :





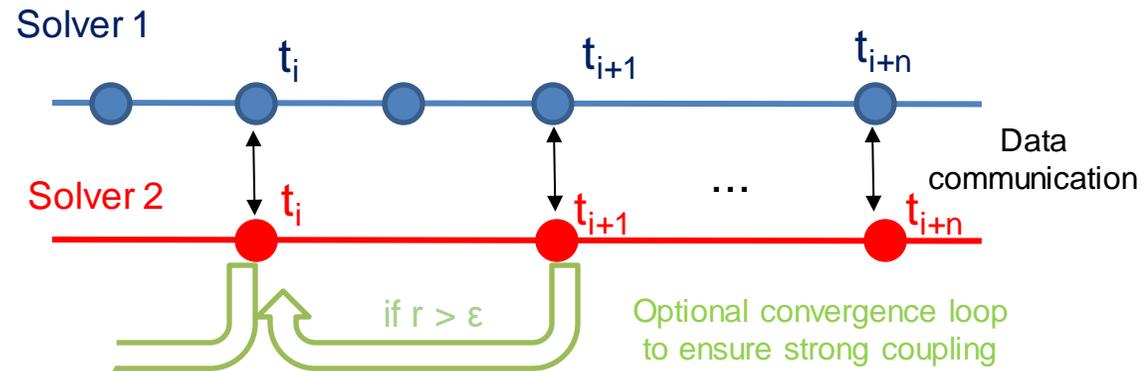
CHESS Numerical Program for Multi-Solver Coupling



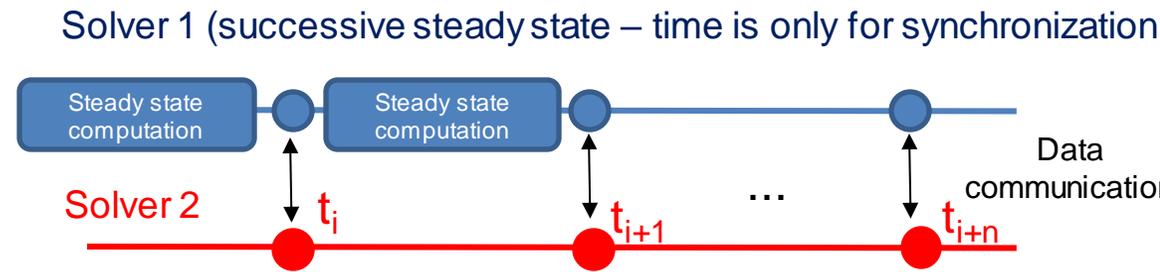
Objective :

- Exploring effect of material response on aerothermal environment,
- Replicating in silico experimental facility for code validation.

Time accurate simulation :



Steady Simulation

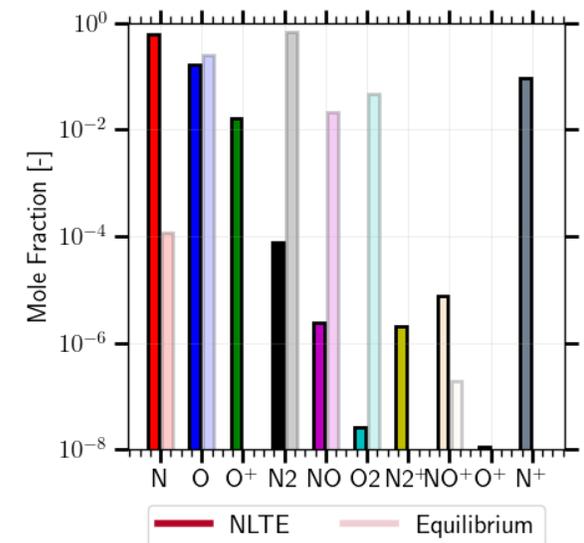
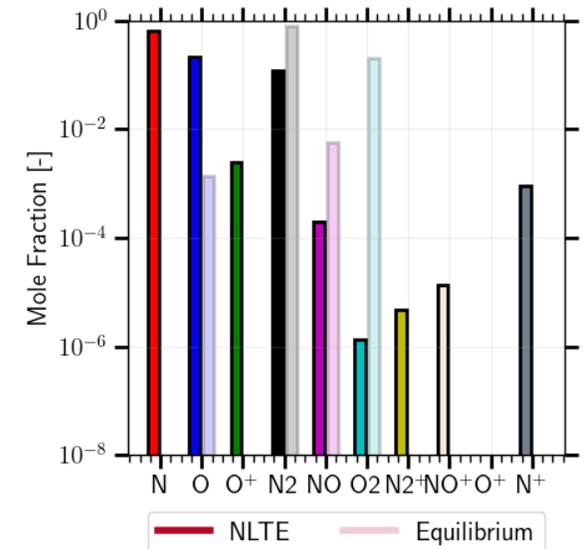
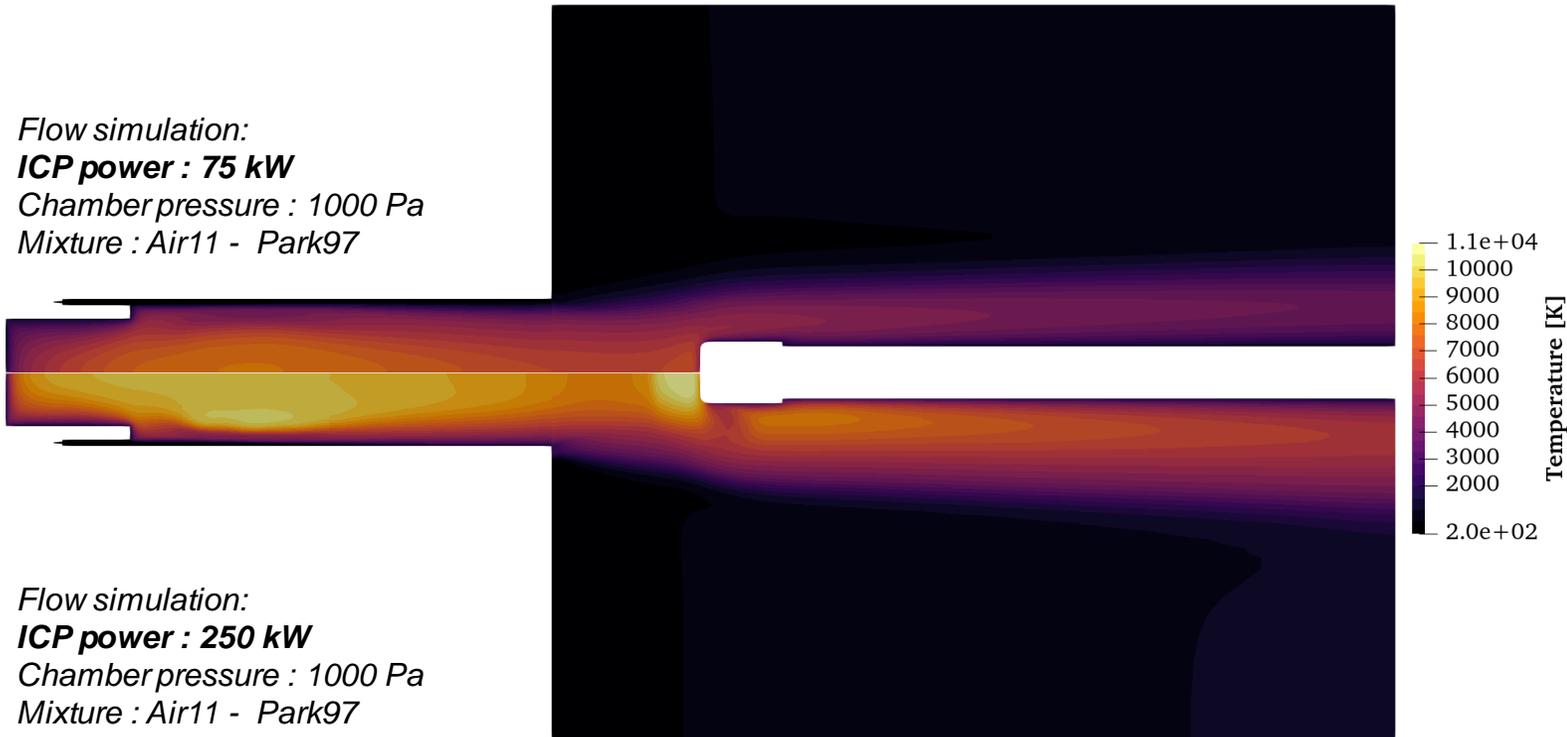


Oxidation in Inductively Coupled Plasma Facility



The numerical experiments have shown the sensitivity of material response on oxidation rate. **How numerical simulations can help to characterize this parameter ?**

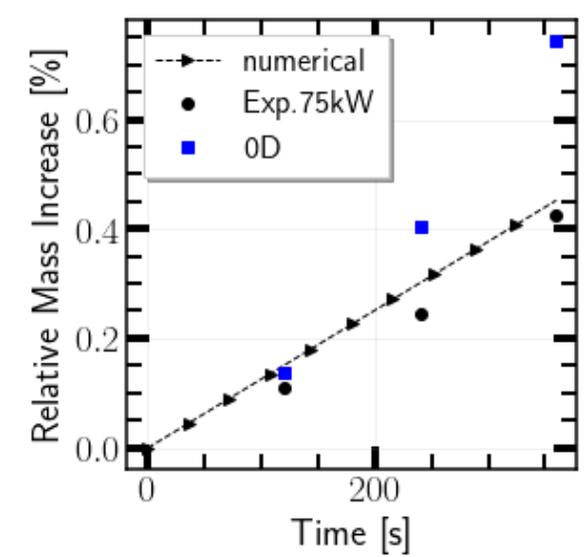
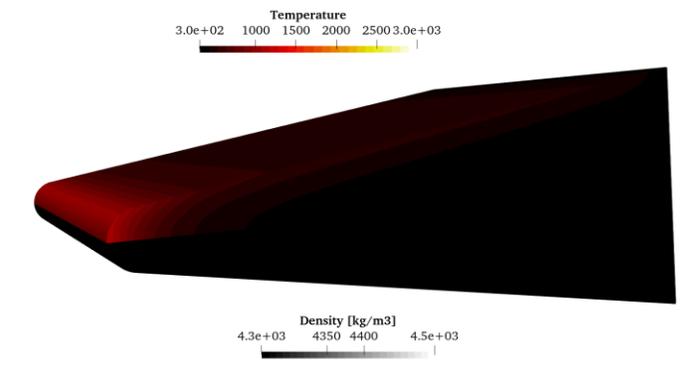
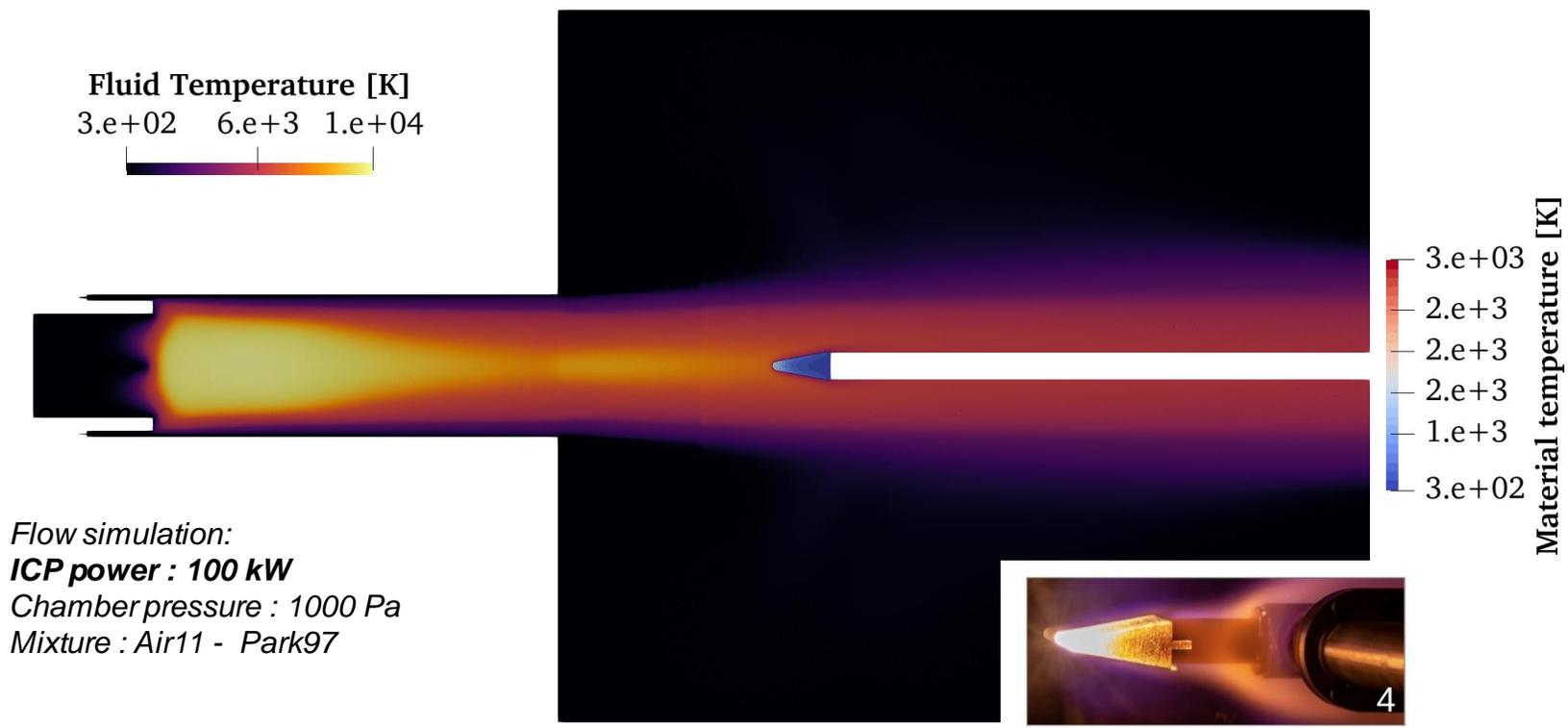
Characterizing aerothermal environment :





Wedge Probe Geometry

The developed framework has then been applied to study material response in ICP flow experiments:





Studies of UHTCs at UIUC as lead to :

- Experimental structural characterization of material (μ CT, XRD surface analysis),
- Oxidation kinetics characterization of ceramics in isobar isothermal furnace experiments,
- Adaptation of ablation module within PATO numerical frameworks for numerical modelling.

Short and mid-term objectives :

- Characterization of UHTC thermo-physical response inside ICP facility,
- Comparison with numerical model developed in UIUC,
- Development of microscale model oxidation model using SPARTA DSMC code to enable multiscale modelling.

