

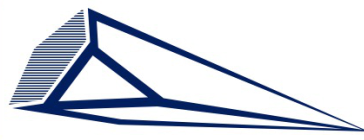


# Numerical Modeling of Leading-Edge Ceramic Oxidation in Inductively Coupled Plasma Facility

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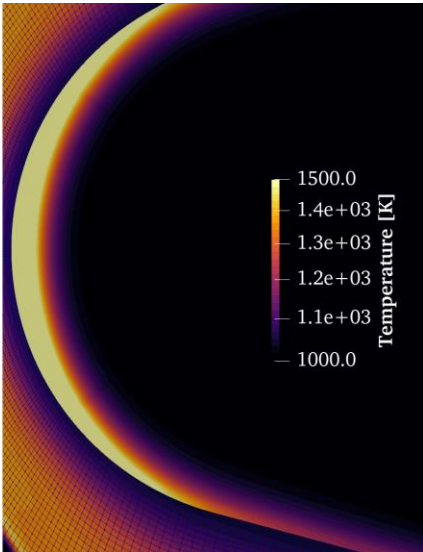
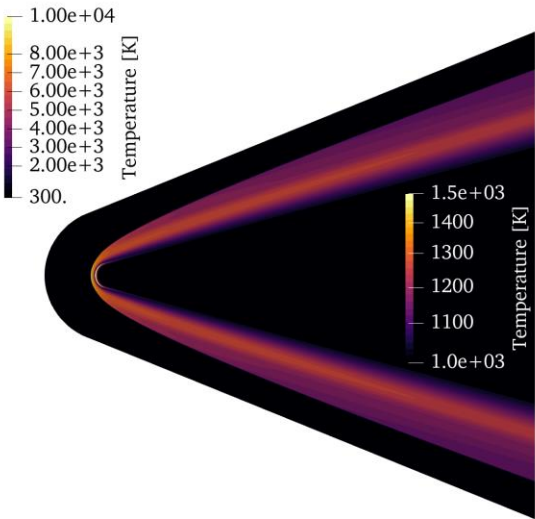


Ablation Workshop  
November 9th-10th 2022

# Ceramics Usage for Hypersonic Applications



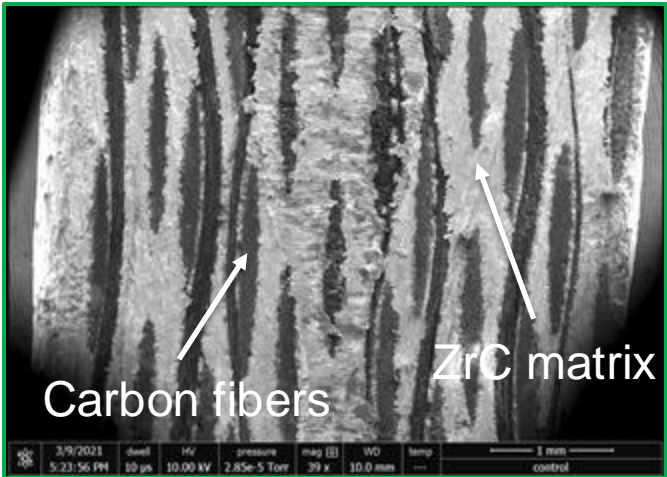
Thermal Protection Systems for atmospheric reentry necessitates high temperature and stress resistance.



Hypersonic reentry simulation at Mach 26.



C/ZrC ceramics composite



Scanning Electron Microscopy

Furnace experiment in an oxygen rich environment



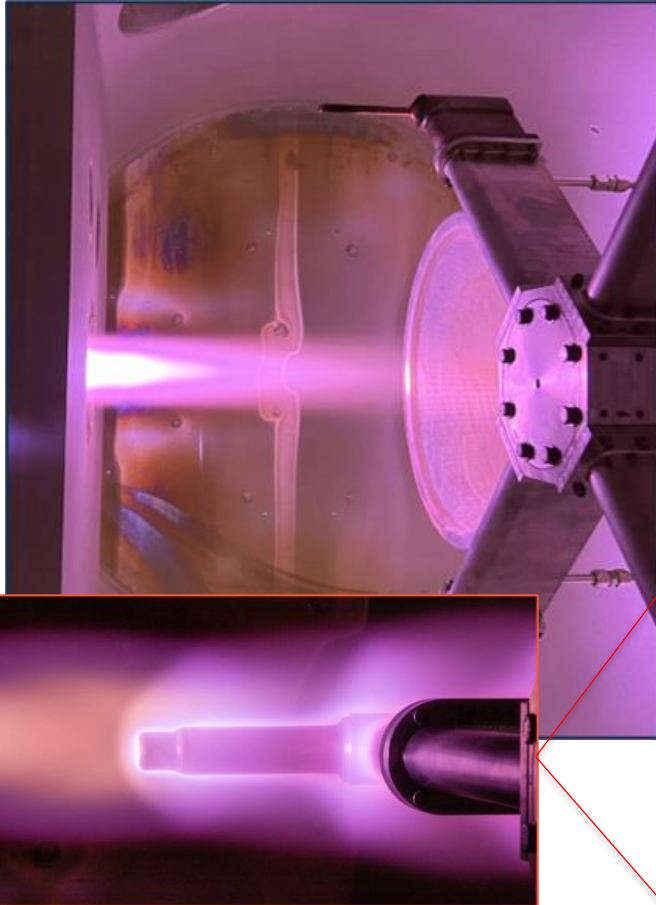
Medium temperature : powdered oxide layer



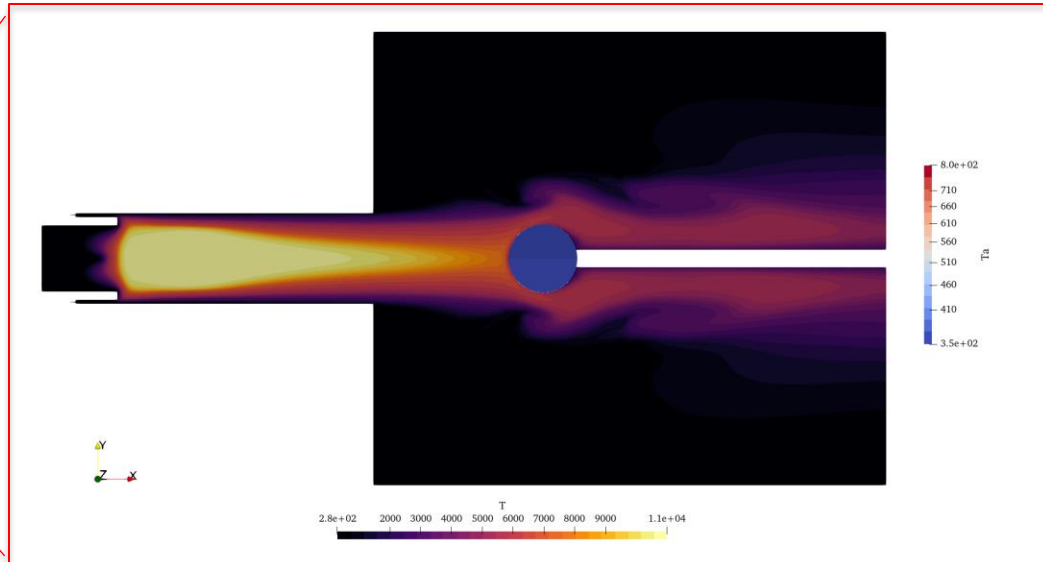
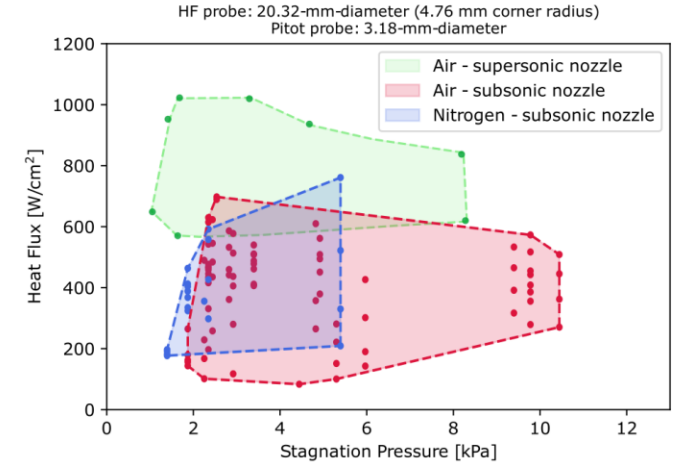
High temperature : dense oxide layer



# Thermal Protection System Testing



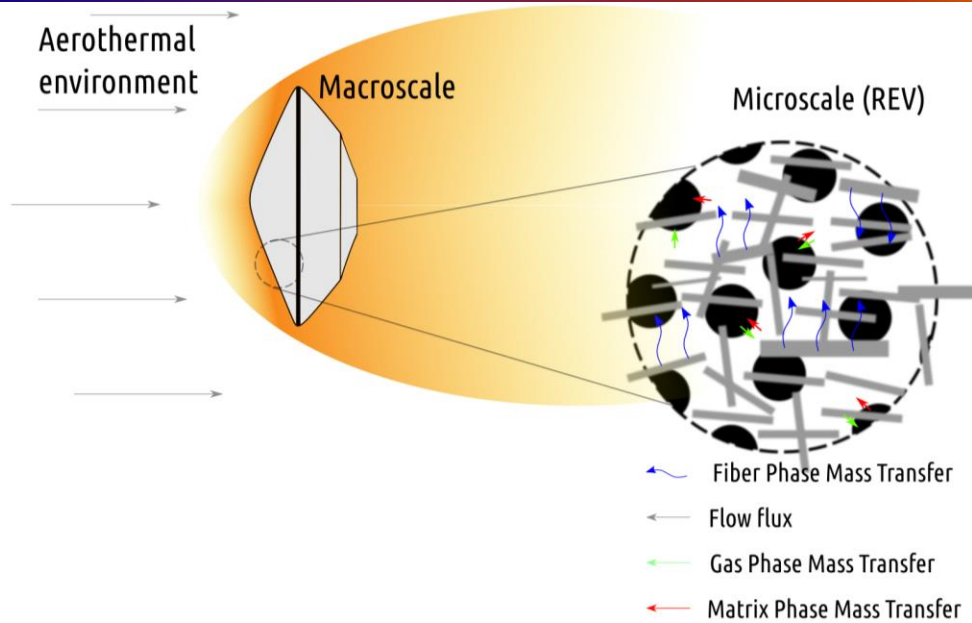
- Inductively Coupled Plasma (ICP) technology
- 350 kW Radio-Frequency Generator (2.1 MHz)
- Maximum performances:
  - Mach 4
  - Stagnation Enthalpies up to 36 MJ/kg
  - Pressure up to 5 bar



Specification overview of some materials testing facilities



# CMC Thermophysical Model



Simulation of Material Response in Aerothermal Environment necessitates :

- Ablation arbitrary number of solid phases,
- Heterogeneous mass exchange,
- Gas species diffusion through the material,
- Global mass and energy conservation.

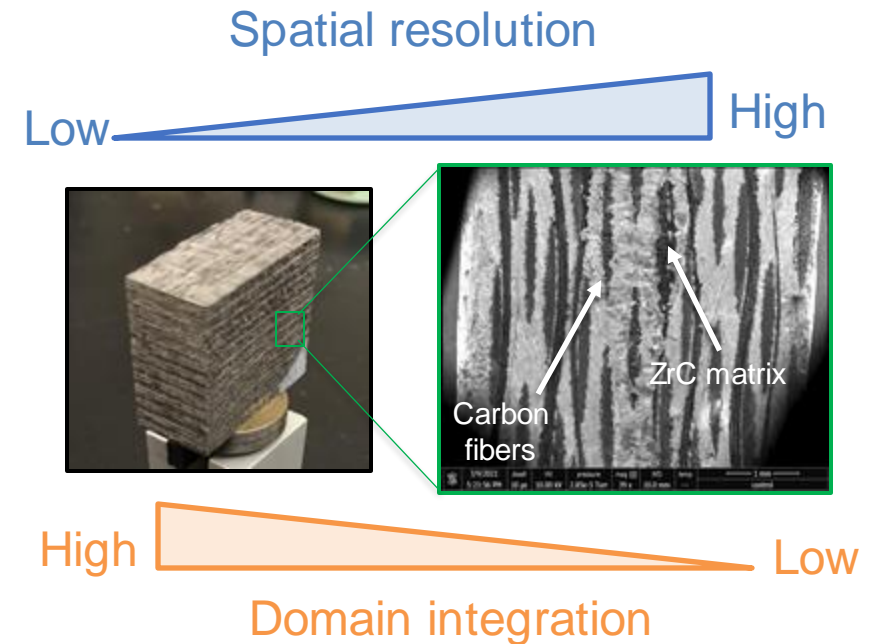
How to leverage the loss of information on large scale modelling ?

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**Thermal Protection Systems** for atmospheric reentry necessitates high temperature and stress resistance.

The **design of protection material** is critical to assess the **safety** of the **payload**.





# Multiscale Description of TPS

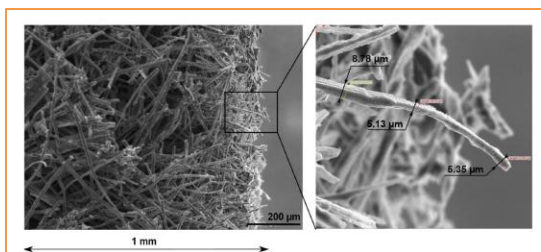


Detailed microscopic mass, momentum and energy balance at the scale for the Representative Elementary Element (REV)

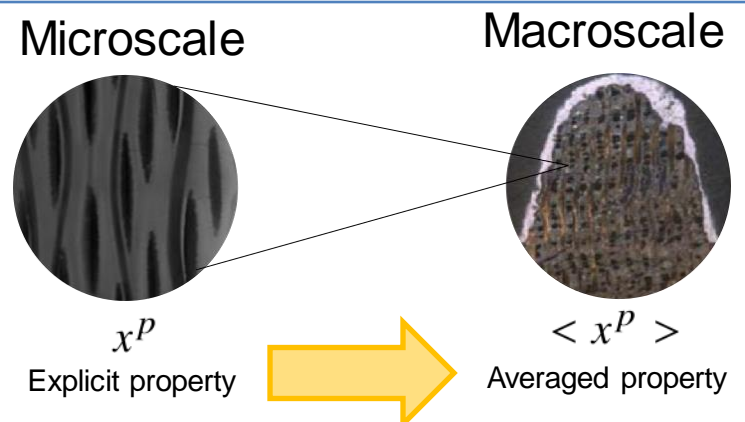
$$\langle x^p \rangle = \frac{1}{V_{REV}} \int_{D^p} x^p dV \quad \forall p \in \mathcal{P}$$

Mathematical Average of microscale governing equations on the REV

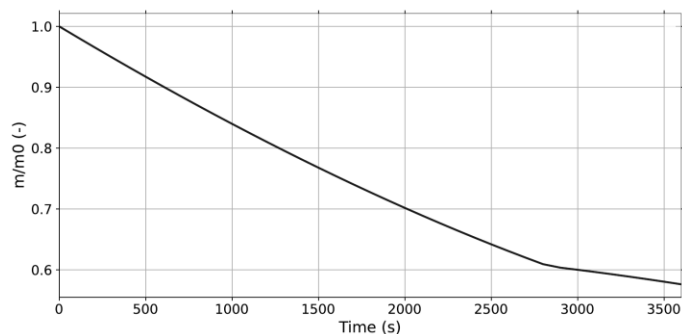
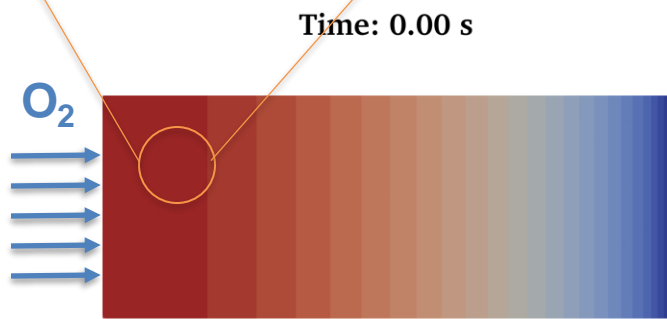
## Multiscale ablation of carbon fibers



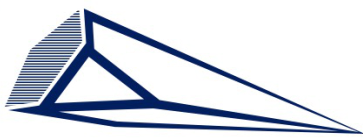
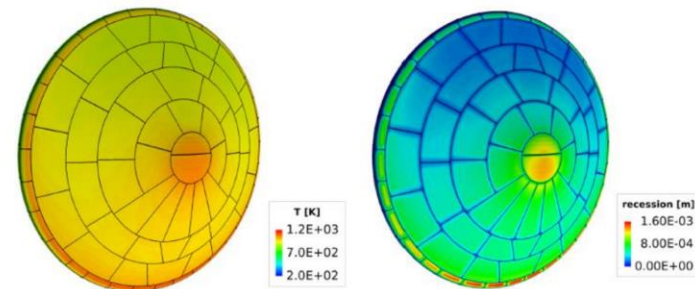
Cylindrical Shrinkage Fiber Model



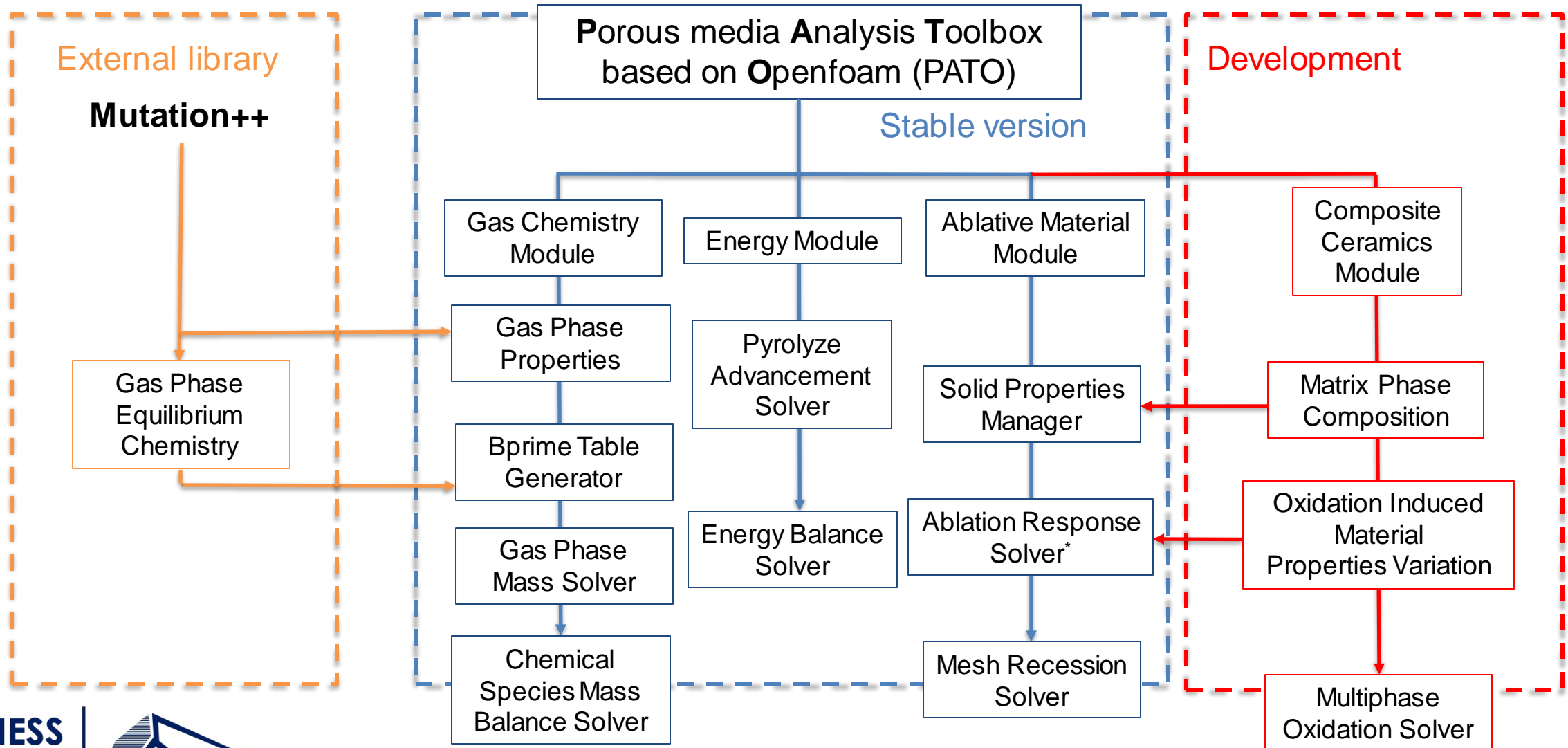
Numerical approximation of resulting PDE on the domain of interest



e.g., Thermochemical simulation of MSL heatshield [1]:



# PATO code overview



# Mathematical Model for Reactive Porous Ceramics



## Mass Conservation in the Gas phase :

- The **perfect gas law** is assumed for the gas phase  $\bar{\rho}^g = \frac{\bar{p}^g RT^g}{M^g}$
- Velocity of gas is computed using the **Darcy-Klinkenberg law**  $\bar{v}^g = -\frac{1}{\epsilon^g} \left( \frac{K^g}{\mu^g} - \frac{\beta}{\bar{p}^g} \right) \nabla \bar{p}^g$
- This allow to define the mass conservation equation of the phase solely as a function of **pressure** :

$$\frac{\partial}{\partial t} \left( \epsilon^g \frac{\bar{p}^g RT^g}{M^g} \right) - \nabla \cdot \left( \frac{\bar{p}^g RT^g}{M^g} \left( \frac{K^g}{\mu^g} - \frac{\beta}{\bar{p}^g} \right) \nabla \bar{p}^g \right) = \underbrace{\sum_{ns=1}^{N_s} \left( \theta^{ns} \sum_k^{I_g} \dot{\omega}_k^{g,ns} \right)}_{\text{Heterogeneous reactions for all solid phases}}$$

- And for gas species k :

$$\frac{\partial}{\partial t} \left( \epsilon^g (\rho^g y_k^g) \bar{g} \right) + \nabla \cdot \left( \epsilon^g (\rho^g y_k^g) \bar{g} \bar{v}^g \right) - \nabla \cdot \left( \underbrace{D_{k \text{ eff}}^g \nabla (\rho^g y_k^g) \bar{g}}_{\text{Diffusion in porous media}} \right) = \underbrace{\epsilon^g \dot{\omega}_k^{\bar{g}}}_{\text{Homogeneous reactions}} + \sum_{ns=1}^{N_s} \underbrace{(\theta^{ns} \dot{\omega}_k^{g,ns})}_{\text{Heterogeneous reactions}}$$



# PATO Oxidation Solver Extension

Accounting for **multiphase oxidations** necessitates to extend **heterogenous reactions solver** to keep track of **gas/surface interactions** and their effect on solid mass balance.

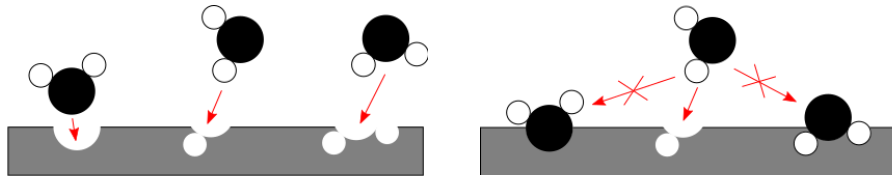
Depths of O<sub>2</sub> penetration is a competitive process between diffusion and surface reactivity measured by the number *Damköhler*:

## Oxidation of CMC

→ O<sub>2</sub> flux    ■ Virgin CMC    ■ Oxidized CMC

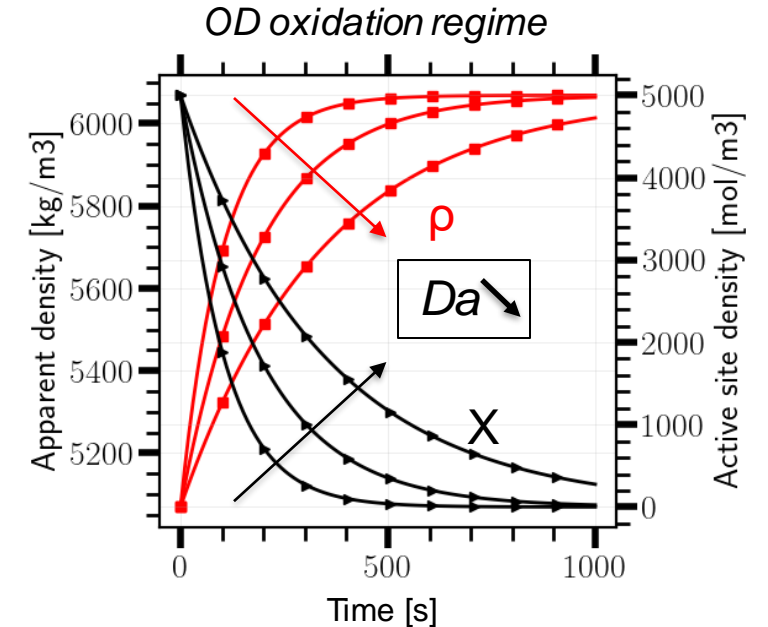
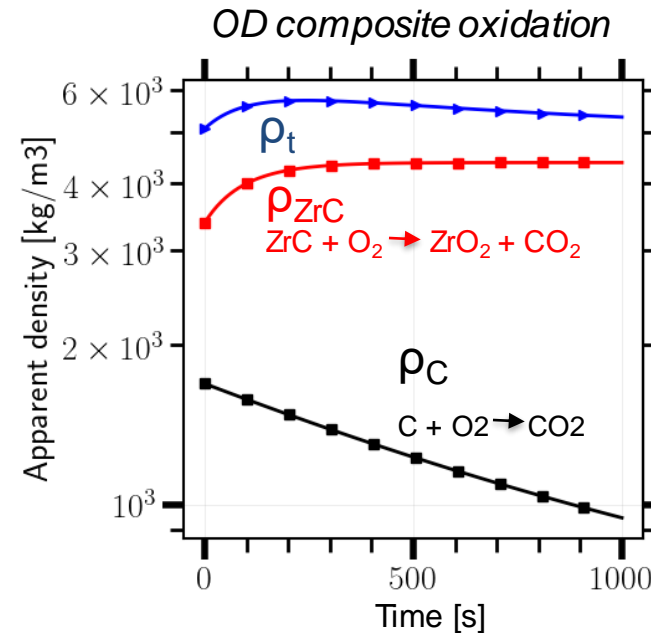


Example : adsorption on a surface



Heterogeneous source term :

$$\frac{\dot{\omega}_k^{g,ns}}{M_k} = \sum_{r=1}^{N_r} \left( \underbrace{k_r^f (\nu_{r,k}'' - \nu_{r,k}') \prod_{j=1}^{I_g \cup I_{ns}} (x_j^{g,ns})^{\nu_{r,j}'}}_{\text{mass production by forward reaction}} - \underbrace{k_r^b (\nu_{r,k}'' - \nu_{r,k}') \prod_{j=1}^{I_g \cup I_{ns}} (x_j^{g,ns})^{\nu_{r,j}''}}_{\text{mass destruction by backward reaction}} \right)$$





## Energy Conservation in the porous media :

- Thermal Equilibrium is supposed to hold in the entire domain  $T^g = T^{ns} \quad \forall ns \in [1, N_s]$
- The conduction parameter is an average of gas and solid phases thermal properties

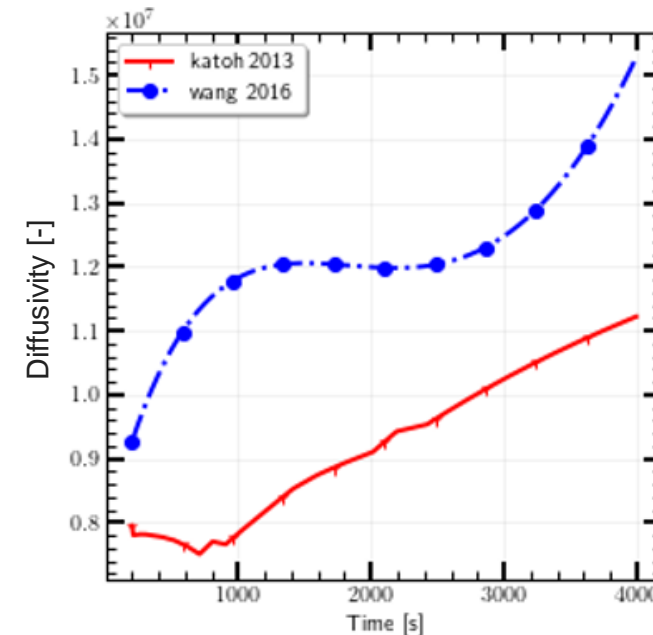
$$\sum_{ns=1}^{N_s} \underbrace{(\epsilon^{ns} \rho^{ns} c_p^{ns})}_{\text{Temperature time variation (heat storage) in solid phases}} \partial_t T + \underbrace{\partial_t (\epsilon^g \rho^g h^g - \epsilon^g p^g)}_{\text{Energy time variation in gas phase}} - \underbrace{\nabla \cdot (\lambda \nabla T)}_{\text{Heat conduction}} + \underbrace{\nabla \cdot (\epsilon^g \rho^g h^g \mathbf{v}^g)}_{\text{Heat convection}} - \underbrace{\nabla \cdot \sum_k \left( D_{k,\text{eff}}^g \nabla (\rho^g y_k^g) \bar{g} h_k^g \right)}_{\text{Heat diffusion}} = \underbrace{\dot{E}_\Omega + \bar{h} \pi_{tot}}_{\text{Heat source}}$$

## Material and Gas Phase Properties :

- Material parameters are function of pressure, temperature and pyrolysis rate
- Gas transport properties are computed using Mutation++

Wei, X. *et al.*, Zirconium Carbide Produced by Spark Plasma Sintering and Hot Pressing: Densification Kinetics, Grain Growth, and Thermal Properties. *Materials* **2016**, 9, 577.

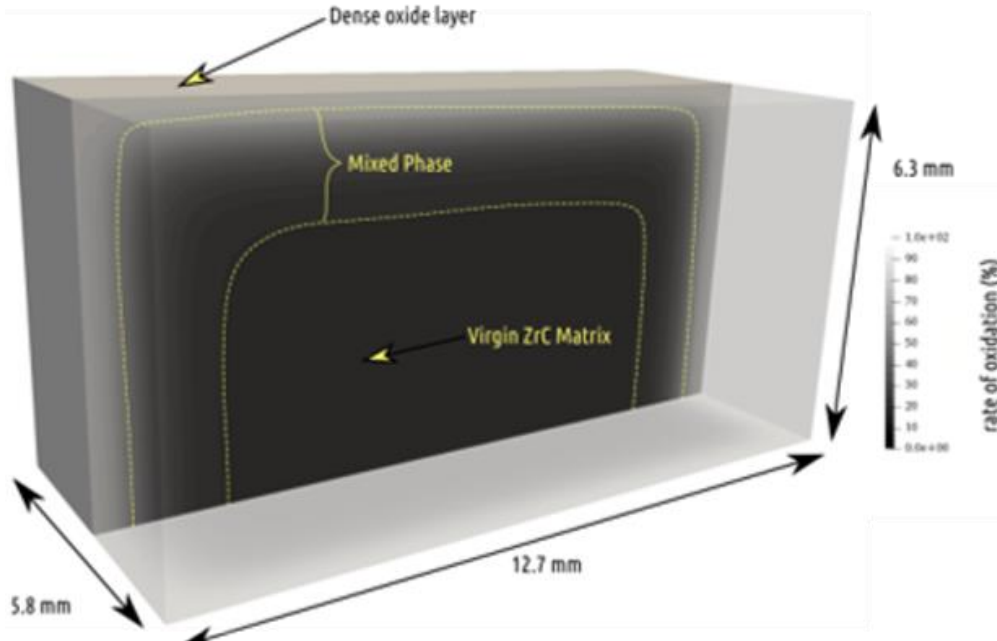
Katoh, Y. *et al.*, Properties of zirconium carbide for nuclear fuel applications. *Journal of Nuclear Materials*, **2013**, 741, 1.



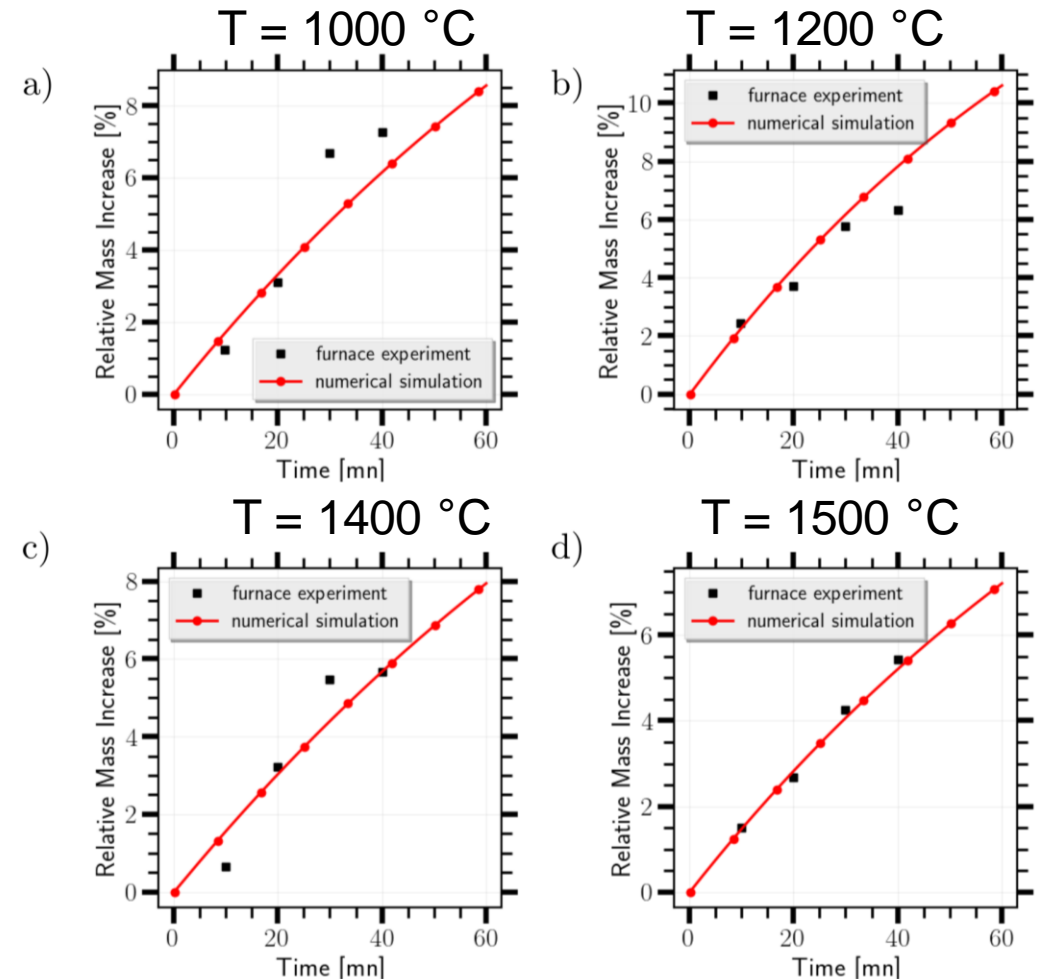
# Multidimensional Oxidation

The model is applied to predict the mass variation of pure ZrC ceramics exposed to constant high temperature at atmospheric pressure environment.

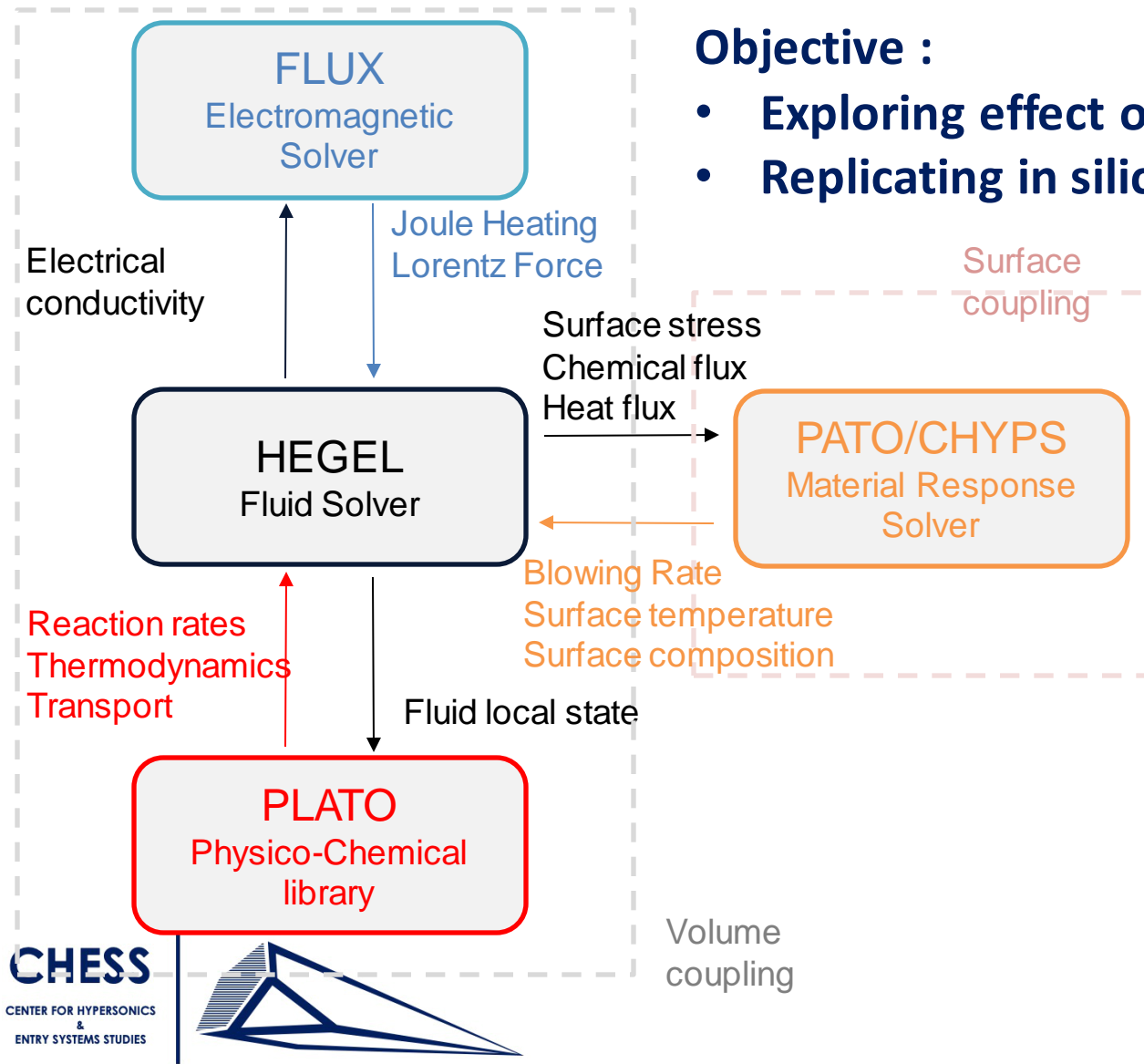
Oxidation of ZrC ceramics at 1400 °C after 20mn



Kinetics rate are obtained from re-characterization of Rama Rao oxidation :



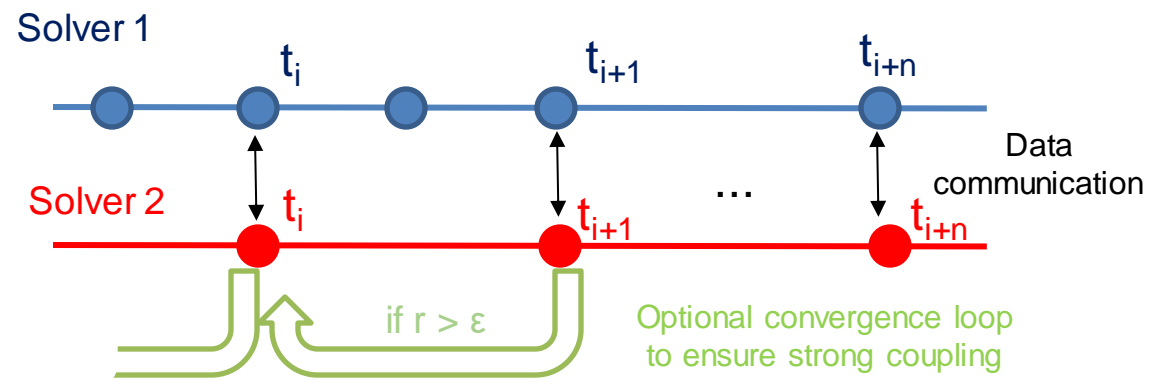
# CHESS Numerical Program for Multi-Solver Coupling



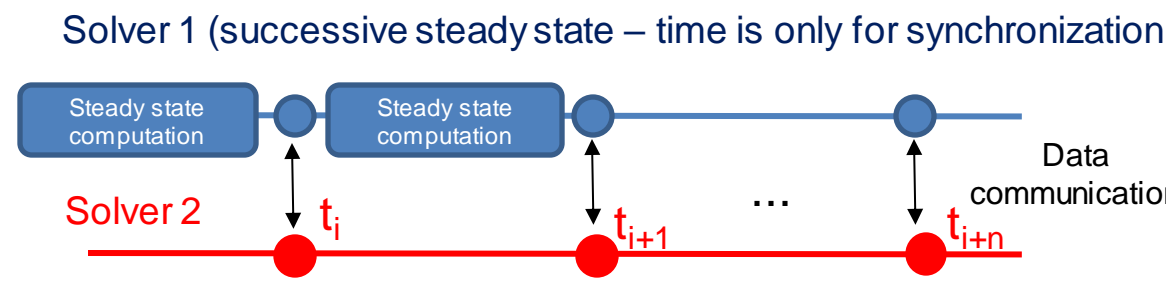
## Objective :

- Exploring effect of material response on aerothermal environment,
- Replicating in silico experimental facility for code validation.

## Time accurate simulation :



## Steady Simulation



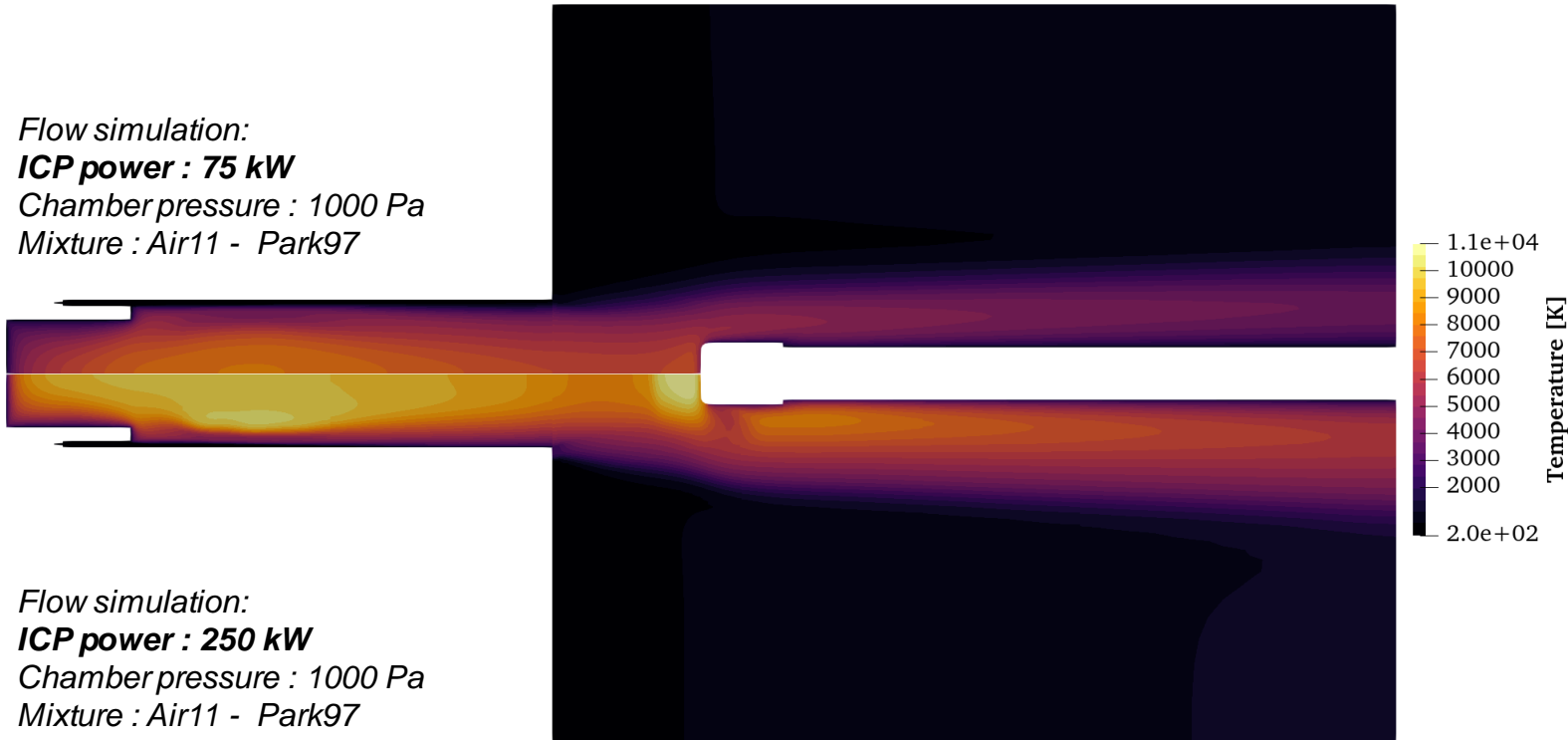
# Oxidation in Inductively Coupled Plasma Facility



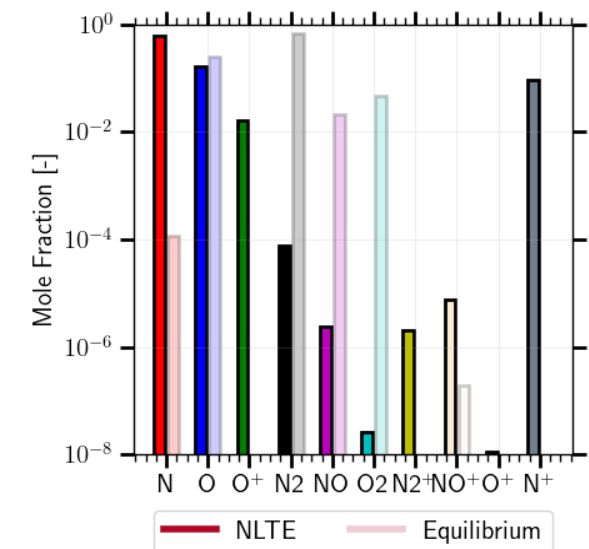
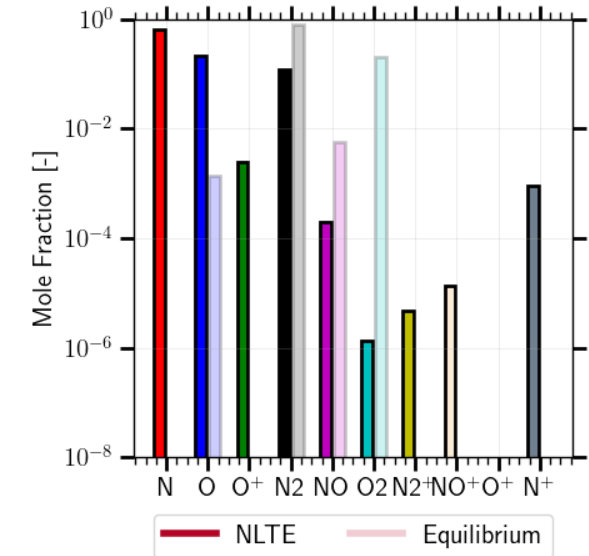
The numerical experiments have shown the sensitivity of material response on oxidation rate. **How numerical simulations can help to characterize this parameter ?**

*Characterizing aerothermal environment:*

Flow simulation:  
**ICP power : 75 kW**  
Chamber pressure : 1000 Pa  
Mixture : Air11 - Park97



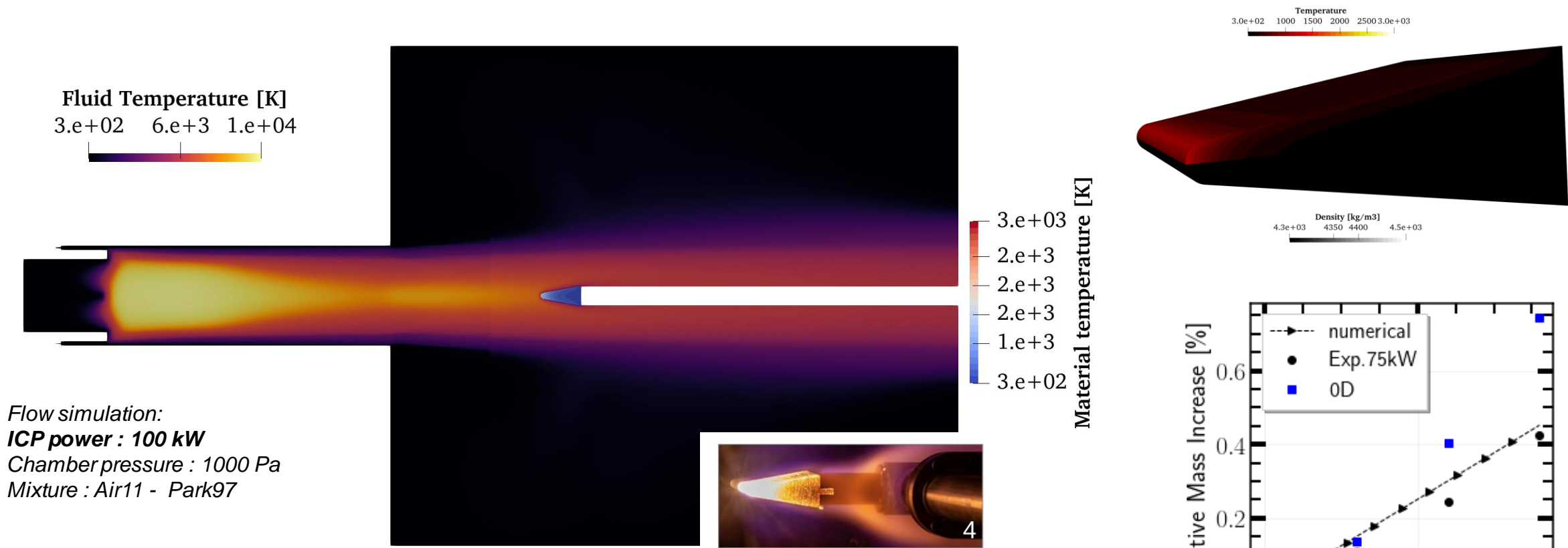
Flow simulation:  
**ICP power : 250 kW**  
Chamber pressure : 1000 Pa  
Mixture : Air11 - Park97



# Wedge Probe Geometry



The developed framework has then been applied to study material response in ICP flow experiments:





## **Studies of UHTCs at UIUC as lead to :**

- Experimental structural characterization of material ( $\mu$ CT, XRD surface analysis),
- Oxidation kinetics characterization of ceramics in isobar isothermal furnace experiments,
- Adaptation of ablation module within PATO numerical frameworks for numerical modelling.

## **Short and mid-term objectives :**

- Characterization of UHTC thermo-physical response inside ICP facility,
- Comparison with numerical model developed in UIUC,
- Development of microscale model oxidation model using SPARTA DSMC code to enable multiscale modelling.

